

Fig. 1. Shapes of the LE symmetrical sections.

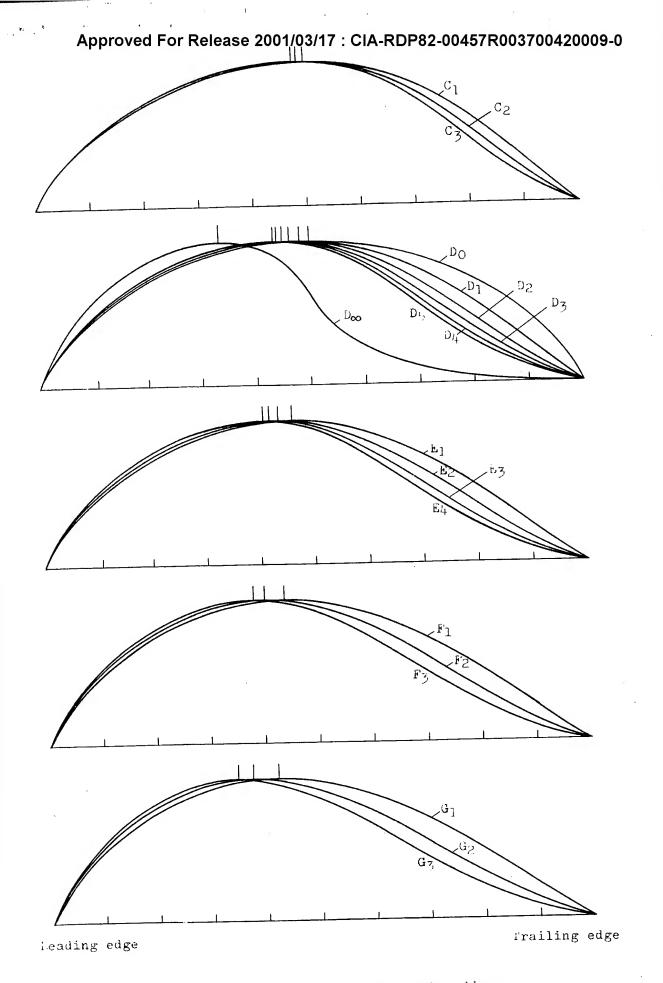
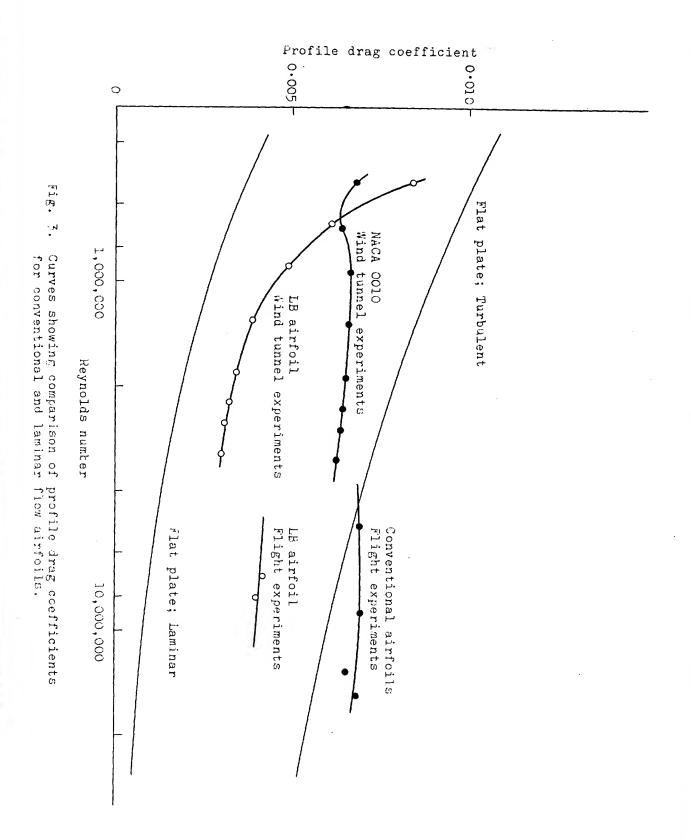
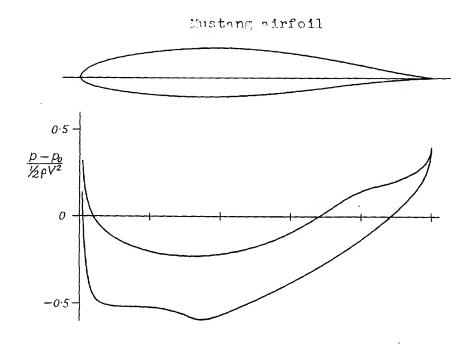


Fig. 2. Shapes of the LB mean camber lines.

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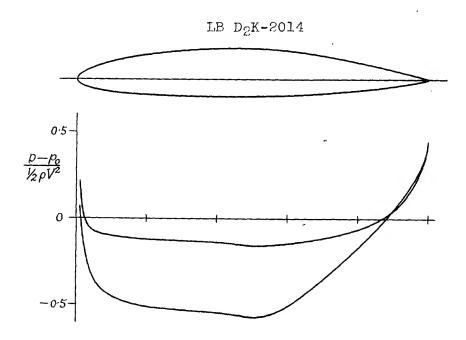


Fig. 4. Curves showing comparison of pressure distributions on the Mustang and LB airfoils at a lift coefficient 0.27.